

GPS-Aided Inertial Navigation System





- Multi frequency and Multi constellations GNSS Receiver
- GPS, GLONASS, GALILEO, QZSS and BEIDOU GNSS
- Designed for fixed-wing and multi-rotor Aerial Platforms
- Two Barometers for Indicated Airspeed
- Position accuracy = 1 cm (RTK)
- Heading accuracy = 0.3 deg (RTK, Dynamic)
- Pitch & Roll accuracy = 0.1 deg (RTK, Dynamic)
- Gyro-compensated, embedded Fluxgate magnetic compass
- Optional external Stand-Alone Magnetic Compass
- Small Size, light weight, low power
- Affordable price







The **Inertial Labs GPS-Aided Inertial Navigation System (INS-U)** is an IP67 rated version of the new generation, fully-integrated, combined Inertial Navigation System (INS) + Attitude & Heading Reference System (AHRS) + Air Data Computer (ADC) high-performance strapdown system, that determines position, velocity and absolute orientation (Heading, Pitch and Roll) for any device on which it is mounted. Horizontal and Vertical Position, Velocity and Orientation are determined with high accuracy for both motionless and dynamic applications.



The Inertial Labs **INS-U** utilizes advanced single antenna multi constellation (GPS, GLONASS, GALILEO, QZSS and BEIDOU GNSS) GNSS receiver; two barometers; miniature gyro-compensated Fluxgate compass; 3-axes each of calibrated in full operational temperature range Advanced MEMS Accelerometers and Gyroscopes to provide accurate Position, Velocity, Heading, Pitch and Roll of the device under measure.

INS-U contains Inertial Labs new on-board sensor fusion filter, state of the art navigation and guidance algorithms and calibration software.

KEY FEATURES, BENEFITS & FUNCTIONALITY

- Commercially exportable GPS-Aided Inertial Navigation System
- 3-in-1 strapdown system: INS + AHRS + ADC (Air Data Computer)
- Embedded in-flight calibration
- Designed for UAV application algorithm
- UBlox ZED-F9P F9 High Precision GNSS Module
- Small size, lightweight & low power: 82 x 40.0 x 26.0 mm, <200-grams, <1 watt
- GPS, GLONASS, GALILEO, BEIDOU, QZSS, RTK supported signals
- Total and Static Pressure Sensors for calculating Indicated Airspeed
- Embedded Gyro-compensated Mini-Fluxgate magnetometers (compass)
- GNSS measurements and IMU raw data for post processing
- Advanced, extendable, embedded Kalman Filter based sensor fusion algorithms
- State-of-the-art algorithms for different dynamic motions of Helicopters, and UAV
- Full temperature calibration of all sensing elements
- Environmentally sealed (IP67)
- Aiding data: Wind sensor, Air Speed Sensor, Doppler shift from locator (for long-term GPS denied), External position and External Heading



GPS-Aided INS-U Preliminary Datasheet Revision 1.8

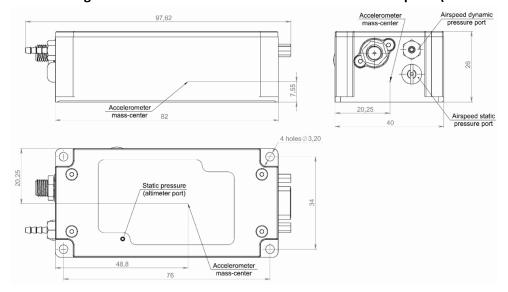
INS-U Specifications

	Specifications -					
	Parameter	Units	INS-U			
	Input signals		External Magnetometer, Wind sensor, Air Speed Sensor, Doppler shift from locator (for long-			
	F 3		term GPS denied), External position and External Heading aiding data			
			IMU data: Accelerations, Angular rates; AUDS data: Magnetic Field Handling Pitch & Ball. AUDS data: Magnetic Field Handling Pitch & Ball. AUDS data: Magnetic Field Handling Pitch & Ball.			
Inputs &			 AHRS data: Magnetic Field, Heading, Pitch & Roll INS data: Positions, Velocity, Delta Theta and Delta Velocity, GNSS data, Time 			
	Output signals		Air Data Computer data: Static Pressure (calibrated), Dynamic Pressure (calibrated), Baro-			
Outputs			Corrected Pressure Altitude, Pressure Altitude, Calibrated Airspeed, True Airspeed, Mach-			
			Number, Static Pressure Over Total Pressure, True Angle of Attack, Rate of Climb			
	Update rate	Hz	1 200 (user settable)			
	Start-up time	sec	<1 <1			
	Positions, Velocity, and Timestamps	Units	INS-U			
	Horizontal position accuracy (SP), RMS	meters	1.5 CEP			
Navigation	Horizontal position accuracy (RTK), RMS (1)	meters	0.01 + 1 ppm CEP			
Navigation	Vertical position accuracy (RTK) (1), RMS	meters	0.01 + 1 ppm CEP			
	Velocity accuracy, RMS	meters/sec	0.05			
	Heading	Units	INS-U			
	Range	deg	0 to 360			
	Angular Resolution	Deg	0.01			
	Static Accuracy (3)	Deg, 1σ	0.3			
Ovientation	Dynamic accuracy (GNSS) (4)	deg RMS, 1σ Units	0.6 INS-U			
Orientation	Pitch and Roll Range: Pitch, Roll	Deg	±90, ±180			
	Angular Resolution	Deg	0.01			
	Static Accuracy in Temperature Range	deg, 1σ	0.05			
	Dynamic Accuracy (4)	deg RMS, 1σ	0.08			
	GNSS receiver	Units	INS-U			
	Туре		Single GNSS Antenna			
	Supported GNSS signals & corrections		GPS L1C/A L2C, GLO L1OF L2OF, GAL E1B/C E5b, BDS B1I B2I, QZSS L1C/A L2C SBAS L1C/A:			
	., .		WAAS, EGNOS, MSAS, GAGAN			
	Channel configuration		184 Channels – F9 Engine			
GNSS	Raw GNSS data rate	Hz	20 (GPS+GLO+GAL+BDS or GPS+GLO+GAL)			
Cition	Assume of Time Dules Cinnel		25 (GPS only or GPS+GAL/GLO/BDS)			
	Accuracy of Time Pulse Signal Frequency of Time Pulse Signal	Ns Hz	30 (RMS), 60 (99%) 0.25 – 10,000 (configurable)			
	GNSS Convergence time (6)	Sec	< 10 (GPS+GLO/GAL/BDS); < 30 (GPS)			
	Acquisition time (7)	Sec	<30 (cold start), <2 (warm start), <1 (hot start)			
	Air Data Computer	Units	INS-U			
	Static Pressure (calibrated)	hPa, % FS	300 to 1100 hPa, from -2000 ft to 30000 ft, Accuracy: ±0.1% FSS			
	Dynamic Pressure (calibrated)	hPa, % FS	0.15 to 25 hPa / 10 to 124 KCAS (600 KCAS is optional), Accuracy: ±0.25% FSS			
	Baro-Corrected Pressure Altitude	meters	-500 to 9000 meters; Accuracy: 1			
	Pressure Altitude	meters	-500 to 9000 meters; Accuracy: 1			
	Calibrated Airspeed	meters/sec	5 to 64 meters/sec (310 meters/sec is optional); Accuracy: 0.5			
Air Data	True Airspeed	meters/sec	5 to 64 meters/sec (310 meters/sec is optional); Accuracy: 0.5			
Computer	Mach-Number	M	0.01 to 0.2 M, Accuracy: 0.001 M			
Computer	Static Pressure Over Total Pressure		0.97 to 1, Resolution 0.000001			
	True Angle of Attack	deg	-50 to 50 deg; Accuracy ±0.25			
	Rate of Climb	meters/sec	±515 meters/sec; Accuracy 0.05			
	Air Density Outside Air Temperature (OAT)	kg/m³ degC	0.3 to 1.6 kg/m³; Accuracy 0.002			
	Wind Speed	meters/sec	-40 to +85 degC; Resolution 0.01 ±200 meters/sec; Accuracy: 0.1			
	Gyroscopes	Units	±200 meters/sec, Accuracy: 0.1 INS-U			
	Measurement range	deg/sec	±2000			
	Bias in-run stability (RMS, Allan Variance)	deg/hr, 1σ	2			
	Angular Random Walk (ARW)	deg/√hr, 1σ	0.38			
	Accelerometers	Units	INS-U			
	Measurement range	g	±8, ±15, ±40			
IMU	Bias in-run stability (RMS, Allan Variance)	mg, 1σ	0.01, 0.03, 0.05			
1110	Velocity Random Walk (VRW)	m/sec/√hr, 1σ	0.02, 0.045, 0.06			
	Magnetometers (embedded)		INS-U			
	Measurement range	Gauss, 1σ	±8.0			
	Bias in-run stability (Allan Variance)	μGauss, 1σ	8			
	Power Spectral Density	μGauss/√Hz, 1σ	15			
	SF Accuracy	%, 1σ	0.05			
	Environment Operating Altitude	Units	INS-U			
	Operating Altitude Humidity	meters %	Up to 10000 meters / 32800 ft <95			
	Operating temperature	deg C	-40 to +85			
	Storage temperature	deg C	-50 to +90			
	Type of Sealing	9 ~	IP-67			
	Sand, Dust, Water, Humidity, Shock, Vibration		MIL-STD-810G			
	MTBF (GM)	hours	100,000			
General	Electrical	Units	INS-U			
General	Supply voltage	V DC	5-32			
	Power consumption	Watts	<1			
	Output Interface (options)	-	RS-232			
	Output data format	-	Binary, NMEA 0183 ASCII characters			
	1 PPS Level	V DC	5 INS-U			
	Physical Nominal Size	Units	82.0 x 40 x 26			
	Weight	mm gram	< 200			
	Weight	grain	< 200			

⁽¹⁾ Measured using 1 km baseline and patch antennas with good ground planes. Does not account for possible antenna phase center offset errors. ppm limited to baselines up to 20 km
(2) RNS, incremental error growth from steady state accuracy. Post-processing results using third party software; (3) in homogeneous magnetic environment, for latitude up to ±65 deg; calibrated in whole operational temperature range; (9) 50% (2) 30 m/s dynamic operation, accuracy may depend on type of motion; (6) depends on atmospheric conditions, baseline length, GNSS antenna, multipath conditions, satellite visibility and geometry. (7) Commanded start. All satellites at – 130 dBm; 1 or specifications are manufactured to a controlled 3 or standard.



INS-U with embedded magnetometer mechanical & electrical interfaces description (default version)



INS-U Part Numbers Structure

Ī	Model	Gyro	Accelerometers	Calibration	Connector	Pressure	Color	Stand Alone	GNSS	Version	Interface
						Ports		Magnetic Compass	receiver		
	INS-U	G2000	A8	TMGA	C15	2P	В	SAMC	ZF9P	V1	.13
			A15								_
			A40								

Example: INS-U-G2000-A15-TMGA-C15-2P-B-ZF9P-V1.13

- INS-U: Enclosed IP67 Rated Version of the Single Antenna GPS-Aided Inertial Navigation System
- G2000: Gyroscopes measurement range = ±2000 deg/sec
- A8: Accelerometers measurement range ±8 g
- A15: Accelerometers measurement range ±15 g
- A40: Accelerometers measurement range ±40 g
- TMGA: Calibration of IMU (Gyroscopes, Accelerometers and Magnetometers) in operational temperature range
- C15: 15 pin micro-D-SUB plug MM-212-015-11 (by Airborn)
- 2P: Two Airspeed Pressure Ports (Total/Static)
- B: Black Color (default)
- SAMC Support external, Stand-Alone Magnetic Compass (optional)
- ZF9P: uBlox ZED-F9P, Dual-Frequency, Multi-Constellation, RTK Capable GNSS Receiver
- V1: Version 1
- VX.13: RS-232/485 (RS-485 for stand-alone magnetic compass only)